

REPORT DOCUMENTATION PAGE		Form Approved OMB No. 0704-0188	
<p>Public reporting burden for the collection of information is estimated to average 1 hour per response, indicating the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and collecting and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of the collection of information, indicating suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0183), Washington, D.C. 20503.</p>			
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE	3. REPORT TYPE AND DATES COVERED	
	February 28, 1997	Final Report: 15 December 1994 through 14 December 1996	
4. TITLE AND SUBTITLE		5. FUNDING NUMBERS	
Integrated Experimental System for Turbulent Flow-Elastic Surface Interactions		AFOSR F49620-95-1-0102	
6. AUTHOR(S)			
Professor Donald Rockwell			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)		8. PERFORMING ORGANIZATION NUMBER	
Department of Mechanical Engineering and Mechanics Lehigh University 354 Packard Laboratory, 19 Memorial Drive West Bethlehem, Pennsylvania 18015		AFOSR-TR-97 G197	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)		SPONSORING/MONITORING AGENCY REPORT NUMBER	
Air Force Office of Scientific Research (AFMC) 110 Duncan Avenue, Suite B115 Bolling Air Force Base, D. C. 20332-0001		<i>DA</i>	
11. SUPPLEMENTARY NOTES			
Imaging, Laser Diagnostics			
12a. DISTRIBUTION/AVAILABILITY STATEMENT		12b. DISTRIBUTION CODE	
Approved for public release; distribution unlimited.			
13. ABSTRACT (Maximum 200 words)			
<p>The overall goal of this investigation is to design, acquire and implement components that form an integrated system for detecting the instantaneous structure of unsteady and turbulent flows, in conjunction with elastic motions of an adjacent surface. Central to this technique is use of a high-powered Argon-ion laser that scans through space at time scales orders of magnitude smaller than the characteristic time scale of the flow. This approach provides the basis for two-dimensional and three-dimensional scans of complex flows. The instantaneous deflection of the surface is to be acquired by a video zoom technique or, alternately, a speckle technique. Quantitative evaluation of the images acquired using this approach involve the use of critical point theory, in conjunction with flow topological techniques and pressure source representations of the unsteady flow. These techniques require the instantaneous, global features of the flow.</p>			
14. SUBJECT TERMS		15. NUMBER OF PAGES	
Imaging, Laser Diagnostics		6	
16. Price code			
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIFICATION OF ABSTRACT	20. LIMITATION OF ABSTRACT
Unclassified	Unclassified	Unclassified	

FINAL TECHNICAL REPORT FOR AFOSR GRANT

INTEGRATED EXPERIMENTAL SYSTEM FOR TURBULENT FLOW - ELASTIC SURFACE INTERACTIONS

P.I. Name: Rockwell, Donald O.

Institution: Lehigh University, 354 Packard Laboratory, 19 Memorial Drive West,
Bethlehem, PA 18015

Contract/Grant No: Equipment Grant F49620-95-1-0102

Effective Dates: 15 December 1994 through 14 December 1996

1. ABSTRACT

The overall goal of this investigation is to design, acquire and implement components that form an integrated system for detecting the instantaneous structure of unsteady and turbulent flows, in conjunction with elastic motions of an adjacent surface. Central to this technique is use of a high-powered Argon-ion laser that scans through space at time scales orders of magnitude smaller than the characteristic time scale of the flow. This approach provides the basis for two-dimensional and three-dimensional scans of complex flows. The instantaneous deflection of the surface is to be acquired by a video zoom technique or, alternately, a speckle technique. Quantitative evaluation of the images acquired using this approach involve the use of critical point theory, in conjunction with flow topological techniques and pressure source representations of the unsteady flow. These techniques require the instantaneous, global features of the flow.

2. EXPERIMENTAL SYSTEMS

A new high-powered Argon continuous wave laser (25 watt rating) has been acquired and interfaced with beam conditioning optics. This optics arrangement combines several types of lenses, including a singlet, in order to provide a sufficiently narrow waist of the beam for impingement upon the rotating multi-faceted mirror, which provides the scanning, and a minimal diameter of the beam at the cross-section of the plane of the flow.

Scanning of the laser beam is achieved by one of two techniques. The first, employing relatively high speeds, involves a multi-faceted rotating mirror. It provides sweeping frequencies up to 2,000 cycles/sec. For lower speed flows, an oscillating, galvanometer-driven mirror is used to provide scan frequencies up to 100 cycles/sec. The three-dimensional scanning version of this system has been designed, and the main housing manufactured. In essence, it employs two orthogonally-oriented rotating mirrors, each having 72 facets, which will provide rapid scanning of the laser beam through three-dimensional space.

Images are recorded using two basic types of film-based systems. The first involves a motor-driven 35 mm camera, having a framing rate of five to ten frames per second. This second is a high-speed framing camera, with a framing rate of up to sixty-five frames per

19970604
141

second. Both of these cameras are synchronized with the laboratory computer system, in order to provide coordination between the camera shutter opening and defined motion of the body or of the flow past a body. This film-based approach has a resolution substantially exceeding that of traditional video systems. High resolution film has an equivalent resolution of 350 pixels/mm; these images are digitized at an effective resolution of 125 pixels/mm.

Video-based image acquisition will involve a newly-released high resolution video camera made by Kodak (model ES.1). In conjunction with an imaging technology incorporated framegrabber, having an effective transfer rate of 90 Megahertz. These images will be downloaded to the ram of the host microcomputer. Since the date is already in digital form, the process of image digitizing will not be necessary. Interrogation of the stored image patterns, to determine the velocity fields will also be done on the host computer. This video system will be interfaced with the rotating mirror. Units are described in the foregoing using a synchronizer developed by TSI, Inc. To effect this interfacing, it will be necessary to employ a Pockels cell system to limit the number of laser beam sweeps for the second range of exposure. By using a cross-correlation between successive video frames, it will be possible to optimize the resolution of the video system, which is rated at approximately 10^6 pixels.

Determination of the surface deflection will be accomplished using one of two techniques. The first involves a highly magnified image of the local region of the surface, which will be tracked using the film-based camera or the video systems described in the foregoing. These local deflections of the surface can be synchronized with the instantaneous, adjacent flow patterns, thereby providing a quantitative relationship between the surface deformation and the predominant features of the flow. This technique has already been implemented for the film-based systems. The second approach involves a speckle technique, for which the video system is ideally suited.

3. ACCOMPLISHMENTS/NEW FINDINGS: POST-PROCESSING OF IMAGES.

The central goal of this instrumentation grant was to relate the underlying physics of the flow to loading and deflection of the adjacent surface. In doing so, several theoretical approaches have been investigated and implemented during the past two years.

Acquisition of instantaneous velocity fields allows determination of the corresponding streamline patterns in an arbitrary frame of reference, thereby allowing definition of the critical points of the flow based on topological techniques. These critical points involve, for example, saddle points, stable and unstable foci, nodal lines, etc. In a number of the investigations cited in the List of References, definition of how these critical points evolve with time has been instrumental in understanding the coupling between surface (body) motion in the surrounding flow.

The instantaneous velocity field will also lead to calculation of pressure source terms, which are important for understanding not only the loading of an adjacent surface, but also the radiated sound from the surface. These source terms have been calculated, and contours of constant pressure sources have been related to corresponding instantaneous

contours of constant vorticity, in order to determine the manner in which the development of the vorticity field gives rise to pronounced pressure sources.

Finally, knowledge of the instantaneous velocity field allows direct calculation of the vorticity. The manner in which this vorticity field evolves with time is central to determining the relation between loading of an oscillating body and the flowfield surrounding it. Two different approaches have been applied. The first involves use of moments of vorticity, in conjunction with the Lighthill concept. The time rate of change of these moments of vorticity is directly related to the instantaneous lift and drag on the body. A second approach, which must be invoked for the case where the vorticity extends well beyond the region of interest of the body, involves a momentum balance, undertaken with aid of the Reynolds Transport Theorem. This approach has proved successful for an oscillating body, provided the reference frame transformations are properly accounted for.

4. PERSONNEL SUPPORTED

Not applicable - this grant was an instrumentation grant with no support provided for personnel.

5. PUBLICATIONS (1995-1997 only)

The experimental systems defined in the foregoing were applied for a broad range of experimental investigations supported by Air Force Office of Scientific Research, the Office of Naval Research, and the National Science Foundation. A summary of the experimental findings is given in each of the publications which benefit from this instrumentation grant.

Journal Articles

Canbazoglu, S., Lin, J.-C. Wolfe, S. and Rockwell, D. 1995 "Buffeting of Fin: Distortion of Incident Vortex", *AIAA Journal*, Vol. 33, No. 11, pp. 2144-2150.

Chyu, C.-K., Lin, J.-C., Sheridan, J. and Rockwell, D. 1995 "Kármán Vortex Formation from a Cylinder: Role of Phase-Locked Kelvin-Helmholtz Vortices", *Physics of Fluids*, Vol. 7, No. 5, pp. 2288-2290.

Cipolla, K. and Rockwell, D. 1995 "Flow Structure on a Stalled Delta Wing Subjected to Small-Amplitude Pitching Motions", *AIAA Journal*, Vol. 33, No. 7, pp. 1256-1262.

Gu, W. and Rockwell, D. 1995 "Flow Structure from an Oscillating Cylinder with a Localized Nonuniformity: Patterns of Coherent Vorticity Concentrations", *Physics of Fluids*, Vol. 7, No. 5, pp. 993-998.

Lin, J.-C., Towfighi, J. and Rockwell, D. 1995 "Instantaneous Structure of Near-Wake of a Circular Cylinder: On the Effect of Reynolds Number", *Journal of Fluids and Structures*, Vol. 9, pp. 409-418.

Lin, J.-C., Towfighi, J. and Rockwell, D. 1995 "Near-Wake of a Circular Cylinder: Control by Steady and Unsteady Surface Injection", *Journal of Fluids and Structures*, Vol. 9, pp. 659-669.

Lin, J. C. And Rockwell, D. 1995 "Evolution of a Quasi-Steady Breaking Wave", *Journal of Fluid Mechanics*, Vol. 302, pp. 29-43.

Lin, J.-C. and Rockwell, D. 1995 "Transient Structure of Vortex Breakdown on a Delta Wing at High Angle-of-Attack", *AIAA Journal*, Vol. 33, No. 1, pp. 6-12.

Lin, J.-C., Vorobieff, P. and Rockwell, D. 1995 "Three-Dimensional Patterns of Streamwise Vorticity in the Turbulent Near-Wake of a Cylinder", *Journal of Fluids and Structures*, Vol. 9, pp. 231-234.

Sheridan, J., Lin, J.-C. and Rockwell, D. 1995 "Metastable States of a Cylinder Wake Adjacent to a Free-Surface", *Physics of Fluids* Vol. 7, No. 9, pp. 2099-2101.

Wolfe, S., Canbazoglu, S., Lin, J.-C. and Rockwell, D. 1995 "Buffeting of Fins: An Assessment of Surface Pressure Loading", *AIAA Journal*, Vol. 33, No. 11, pp. 2232-2235.

Wolfe, S., Lin, J.-C. and Rockwell, D. 1995 "Buffeting of the Leading-Edge of a Flat Plate due to a Streamwise Vortex: Flow Structure and Surface Pressure Loading", *Journal of Fluids and Structures*, Vol. 9, pp. 359-370.

Canbazoglu, S., Lin, J.-C., Wolfe, S. and Rockwell, D. 1996 "Buffeting of a Fin: Streamwise Evolution of Flow Structure", *AIAA Journal of Aircraft*, Vol. 33, #1, pp. 185-190.

Chyu, C.-K. and Rockwell, D. 1996 "Near-Wake Structure of an Oscillating Cylinder: Effect of Controlled Kelvin-Helmholtz Vortices", *Journal of Fluid Mechanics*, Vol. 322, pp. 21-49.

Chyu, C.-K. and Rockwell, D. 1996 "Kármán Vortex Development: Relation to Symmetry of Transition Waves", *AIAA Journal*, Vol. 34, pp. 1954-1956.

Chyu, C.-K. and Rockwell, D. 1996 "Evolution of Streamwise Vorticity and Spanwise Modes in the Turbulent Near-Wake of a Cylinder", *Journal of Fluid Mechanics*, Vol. 320, pp. 117-137.

Jefferies, R. and Rockwell, D. 1996 "Interactions of a Vortex with an Oscillating Leading-Edge", *AIAA Journal*, Vol. 34, No. 11, pp. 2448-2450.

Jefferies, R. and Rockwell, D. 1996 "Oscillations of the Vortex with an Oscillating Leading-Edge", 1996 *AIAA Journal*, Vol. 35, No. 3, pp. 2448-2450.

Lin, J.-C., Sheridan, J. and Rockwell, D. 1996 "Near-Wake of a Perturbed, Horizontal Cylinder at a Free-Surface", *Physics of Fluids*, Vol. 8, No. 8, pp. 2107-2116.

Lin, J.-C., Phetkong, N., Sheridan, J. and Rockwell, D. 1996 "Controlled Motion of a Cylinder Through a Free-Surface: Effect of Depth of Penetration", *Journal of Fluids and Structures*, Vol. 10, No. 4, pp. 309-318.

Lin, J.-C., Vorobieff, P. and Rockwell, D. 1996 "Space-Time Imaging of a Turbulent Near-Wake by High-Image-Density Particle Image Cinematography", *Physics of Fluids*, Vol. 8, No. 2, pp. 555-564.

Lin, J.-C. and Rockwell, D. 1996 "Force Identification by Vorticity Fields: Techniques Based on Flow Imaging", *Journal of Fluids and Structures*, Vol. 10, No. 6, pp. 663-668.

Vorobieff, P. and Rockwell, D. 1996 "Multiple-Actuator Control of Vortex Breakdown on a Pitching Delta Wing", *AIAA Journal*, Vol. 34, No. 10, pp. 2184-21886.

Vorobieff, P. and Rockwell, D. 1996 "Wavelet Filtering for Topological Decomposition of Flowfields", *International Journal of Imaging Systems and Technology*, Vol. 7, pp. 211-214.

Lin, J.-C. and Rockwell, D. 1997 "Quantitative Interpretation of Vortices from a Cylinder Oscillating in Quiescent Fluid", *Experiments in Fluids*, in press.

Sheridan, J., Lin, J.-C. and Rockwell D. 1997 "Flow Past a Cylinder Close to a Free-Surface", *Journal of Fluid Mechanics*, in press.

Akin, O. and Rockwell, D. 1997 "Flow Structure from the trailing-edge of a stalled impeller blade", submitted to *Journal of Fluids and Structures*.

Cipolla, K. M. and Rockwell D. 1997 "Crossflow Topology of Flow Structure on a Delta Wing: Effect of Self-Excited Excursions of Vortex Breakdown", submitted to *AIAA Journal of Aircraft*.

Cipolla, K. M., Liakopoulos, A, and Rockwell, D. 1997 "Quantitative Imaging in Proper Orthogonal Decomposition of Flow Past a Delta Wing", submitted to *AIAA Journal*.

Rockwell, D. 1997 "Three-Dimensional Flow-Structure on Delta Wings at High Angle-of-Attack: Experimental concepts and Issues", submitted to *AIAA Journal*.

Shiang, A., Lin, J.-C., Öztekin, A. and Rockwell, D. 1997 "Viscoelastic Flow Around a Confined Circular Cylinder: Measurements Using High-Image-Density Particle Image Velocimetry", submitted to *Journal of Non-Newtonian Fluid Dynamics*.

Sheridan, J., Lin, J.-C. and Rockwell, D. 1997 "Flow Structure in the Near-Wake of a Partially-Submerged Cylinder. Part I: Mechanisms of Instantaneous Reattachment. Part II: Wake-Free Surface Coupling", to be submitted to *Journal of Fluids and Structures*.

Vorobieff, P. and Rockwell, D. 1997 "Control of Vortex Breakdown on a Pitching Half-Delta Wing by Intermittent Trailing-Edge Blowing", to be submitted to *AIAA Journal*.

Abstracts (APS, 1996)

Cetiner, O., Zhu, Q., Lin, J.-C., Unal, M. F. and Rockwell, D. 1996 "Orbital Oscillations of a Cylinder in Presence of a Free-Surface: Vortex Formation and Loading", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. HD4, p. 1810

Cipolla, K. M., Liakopoulos, A. and Rockwell, D. 1996 "Quantitative Imaging in Proper Orthogonal Decomposition of Flow Past a Delta Wing", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. IE3, p. 1826.

Han, S.-H., Phetkong, N., Lin, J.-C. and Rockwell, D. 1996 "Free-Surface Signatures and Vorticity Patterns Due to a Long Wave Motion Past a Surface Piercing Cylinder", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. DJ3, p. 1732.

Lin, J.-C. and Rockwell, D. 1996 "Unidirectional Oscillations of a Cylinder Beneath a Free-Surface: Interrelationship Between Vortex Formation, Loading, and Free-Surface Distortion", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. HD3, p. 1809.

Oshkai, P., Downes, K., Lin, J.-C. and Rockwell, D. 1996 "A Free-Surface Wave Past an Oscillating Cylinder: Control Concepts", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. HD2, p. 1809, 1996.

Unal, M. F. and Rockwell D. 1996 "Vortex-Vortex Interactions Behind an Oscillating Plate", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. FI8, p. 1779.

Zhu, Q., Cetiner, O., Lin, J.-C., Unal, M.F. and Rockwell, D. 1996 "Impulsive and Periodic Response of a Cylinder-Free Surface System: Vorticity Field and Loading", *Bulletin of the American Physical Society*, Vol. 41, No. 9, Abstract No. HD1, p. 1809.